

## State-Space Representations and Examples

In the usual study of DSP, we describe LTI systems with an impulse response,  $h(n)$ ; frequency response,  $H(e^{j\omega})$ ;

difference equation,  $y(n) = -\sum_{m=1}^M a_m y(n-m) + \sum_{l=0}^L b_l x(n-l)$ ; transfer function,  $H(z)$ ; or a pole/zero pattern.

We also can describe a linear system with a state-variable description or state-space representation. Consider the following example.

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**Example:** Suppose we have a simple second order difference equation

$$y(n) = -a_1 y(n-1) - a_2 y(n-2) + x(n) + b_1 x(n-1) + b_2 x(n-2)$$

and the equivalent Direct Form II (Canonical) realization.

Figure (actual DFII and generalized “state” structure)

To be more general, we denote the internal filter “state” with  $S$  and the state variables with  $q_i(n)$ . The block diagram indicates that the next value of the state is obtained from the current state and the input via the equations

$$q_1(n) = -a_1 q_1(n-1) - a_2 q_2(n-1) + x(n)$$

$$q_2(n) = q_1(n-1)$$

The block diagram also indicates that the system output is expressed in terms of the input and state as

$$y(n) = (b_1 - a_1)q_1(n) + (b_2 - a_2)q_2(n) + x(n).$$

In matrix form we can write a state update equation as

$$\begin{bmatrix} q_1(n) \\ q_2(n) \end{bmatrix} = \begin{bmatrix} -a_1 & -a_2 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} q_1(n-1) \\ q_2(n-1) \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \end{bmatrix} x(n)$$

$$\mathbf{q}(n) = \mathbf{A}\mathbf{q}(n-1) + \mathbf{b}x(n)$$

and the output equation as

$$y(n) = [(b_1 - a_1) \quad (b_2 - a_2)] \begin{bmatrix} q_1(n) \\ q_2(n) \end{bmatrix} + [1]x(n)$$

$$y(n) = \mathbf{c}\mathbf{q}(n) + dx(n)$$

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Using the standard notation for state-space descriptions we can generalize for multiple inputs and multiple outputs (MIMO) and allow for the linear system to be dynamical (time-varying)

$$\begin{array}{l} \mathbf{x}(n) = \mathbf{A}(n-1)\mathbf{x}(n-1) + \mathbf{B}(n)\mathbf{w}(n) \\ \mathbf{y}(n) = \mathbf{C}(n)\mathbf{x}(n) + \mathbf{D}(n)\mathbf{v}(n) \end{array}$$

where

State vector or signal model:  $\mathbf{x}(n) = [x_1(n) \ x_2(n) \ \dots \ x_p(n)]^T$

Output vector or observation model or measurement model:  $\mathbf{y}(n) = [y_1(n) \ y_2(n) \ \dots \ y_q(n)]^T$

Process noise vector or modeling error vector:  $\mathbf{w}(n) = [w_1(n) \ w_2(n) \ \dots \ w_p(n)]^T$

Measurement noise vector or observation error vector:  $\mathbf{v}(n) = [v_1(n) \ v_2(n) \ \dots \ v_q(n)]^T$

State transition matrix,  $\mathbf{A}(n)$  ( $p \times p$ )

Input matrix,  $\mathbf{B}(n)$  ( $p \times 1$ )

Observation matrix or measurement matrix or output matrix,  $\mathbf{C}(n)$  ( $q \times p$ )

Direct transmission matrix,  $\mathbf{D}(n)$  ( $q \times 1$ )

Note that with this new notation,  $\mathbf{w}(n)$  is the input to the signal model and  $\mathbf{x}(n)$  is the input to the observation model

Figure 7.11 (Manolakis)

It should be noted that from the state-space representation we can get alternate descriptions [ $\mathbf{D}$  a scalar (straight-through path)] such as the transfer function,

$$H(z) = \mathbf{D} + \mathbf{C}(z\mathbf{I} - \mathbf{A})^{-1}\mathbf{B}$$

or the impulse response (for causal systems)

$$h(n) = \begin{cases} 0, & n < 0 \\ \mathbf{D}, & k = 0 \\ \mathbf{C}\mathbf{A}^{n-1}\mathbf{B}, & k > 0 \end{cases}$$

or the poles of the system. We have  $A(z) = \det(z\mathbf{I} - \mathbf{A})$  thus the eigenvalues of  $\mathbf{A}$  are the poles. Similar result for zeros.

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**Example** (Manolakis): Suppose we have an AR process described by (not the usual notation from Chapter 2)

$$x(n) = 1.8x(n-1) - 0.81x(n-2) + 0.1w(n)$$

where  $w(n)$  is zero mean, unit variance white Gaussian noise. Suppose we can only make “noisy” measurements of the process

$$y(n) = x(n) + \sqrt{10}v(n)$$

where  $v(n)$  is zero mean, variance of 10 white Gaussian noise independent of  $w(n)$ . We have as our state-space description

$$\begin{cases} \mathbf{x}(n) = \mathbf{A}(n-1)\mathbf{x}(n-1) + \mathbf{B}(n)\mathbf{w}(n) \\ \mathbf{y}(n) = \mathbf{C}(n)\mathbf{x}(n) + \mathbf{D}(n)\mathbf{v}(n) \end{cases}$$

where

$$\mathbf{A}(n) = \begin{bmatrix} 1.8 & -0.81 \\ 1 & 0 \end{bmatrix}, \mathbf{B}(n) = \begin{bmatrix} 0.1 \\ 0 \end{bmatrix}, \mathbf{C}(n) = [1 \quad 0], D = \sqrt{10}.$$

This system is not dynamic (fixed  $\mathbf{A}$ ,  $\mathbf{B}$ ,  $\mathbf{C}$ ,  $\mathbf{D}$ ).

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**Example (De Leon):** Suppose we wish to describe the trajectory of a projectile shot with an initial velocity of 1m/s at an angle of  $\pi/4$  from the horizon.

Figure

Using basic Newtonian laws of motion (discretized) we have functions for distance (range),  $d(n)$  and velocity,  $v(n)$ :

$$d(n) = d(n-1) + v(n-1)T + \frac{1}{2}a(n-1)T^2$$

$$v(n) = v(n-1) + a(n-1)T$$

where  $a(n)$  is the acceleration and  $T$  is the measurement sample period. We can construct two, three-state vectors for each direction (x- and y-):

$$\mathbf{x}_x(n) = \begin{bmatrix} d_x(n) \\ v_x(n) \\ a_x(n) \end{bmatrix}, \mathbf{x}_y(n) = \begin{bmatrix} d_y(n) \\ v_y(n) \\ a_y(n) \end{bmatrix}.$$

In this case the state update equation would be

$$\mathbf{x}(n) = \mathbf{A}(n-1)\mathbf{x}(n-1)$$

where

$$\mathbf{A}(n) = \begin{bmatrix} 1 & T & \frac{1}{2}T^2 \\ 0 & 1 & T \\ 0 & 0 & 1 \end{bmatrix}$$

initialized with

$$\mathbf{x}_x(0) = \begin{bmatrix} 0 \\ \sqrt{2}/2 \\ 0 \end{bmatrix}, \mathbf{x}_y(0) = \begin{bmatrix} 0 \\ \sqrt{2}/2 \\ -9.8 \end{bmatrix}$$

We can track the projectile (or determine its trajectory) with the output equation

$$\mathbf{y}(n) = \mathbf{C}(n)\mathbf{x}(n)$$

where  $\mathbf{C}(n) = [1 \quad 0 \quad 0]$  (masks off velocity and acceleration in state vector). The absence of  $\mathbf{B}(n)$  implies a perfect model of the system while the absence of  $\mathbf{D}(n)$  implies we can make perfect observations (measurements). This system is also not dynamic.